

**In the Abstract**

Please amend the Abstract as presented in the underlying International Application No. PCT/EP2004/003642 as follows:

**ABSTRACT**

A rotor, in particular for a rotorcraft, encompassing: a rotor head, at least one rotor blade (2), and a rotor-head-end rotor-blade connector (4; ~~6, 8~~) having an integral, bearingless, centrifugal-force-controlled blade angle adjustment device (~~6, 8~~). A rotorcraft, in particular a helicopter, in particular a tiltrotor helicopter, encompassing at least one such rotor. A method for adjusting the blade angle of a rotor blade (2) of a rotor, in particular of a bearingless rotor that possesses a rotor head and a rotor-head-end rotor-blade connector, encompassing the following steps: rotating the rotor blade (2); and automatically adjusting the blade angle (~~W~~) by twisting the rotor-head-end rotor-blade connector (~~4; 6, 8~~), and thus the rotor blade (2), about its longitudinal axis by means of centrifugal forces (~~F<sub>c</sub>~~) acting on the rotor blade (~~2~~).